

CLARK Y AIRFOIL TABLE

%	0	2.5	5	10	20	30	40	50	60	70	80	90	100
Upper	3.5	6.5	7.9	9.6	11.4	11.7	11.4	10.5	9.2	7.4	5.2	2.8	0.1
Lower	3.5	1.5	0.9	0.4	0	0	0	0	0	0	0	0	0

WING CHORD = 10"

Upper	.4"	.7"	.8"	1"	1.1"	1.2"	1.1"	1.1"	.9"	.7"	.5"	.3"	.1"
Lower	.4"	.2"	.1"	0"	0"	0"	0"	0"	0"	0"	0"	0"	0"

