

# LAA/MOD 3 MODIFICATION APPLICATION Issue 5

 Mod	No.	

This form is used to describe a modification in detail to constitute part of the aircraft build standard record. All pages must be included in any submission. A completed Modification Proposal form (LAA/MOD2) must have been submitted prior to submitting this form, see  $\underline{\text{TL 3.01}}$ .

This form may be printed out, completed by hand and either posted to LAA Engineering or scanned and emailed to <a href="mailto:engineering@laa.uk.com">engineering@laa.uk.com</a>, or it may be completed electronically, saved and emailed as an attachment to the same email address. If emailed without the owner's signature, it must be sent by the aircraft owner. Please retain a copy of the completed form for your records.

1. AIRCRAFT DETAILS

Reg	gistration	Туре	Serial Number
G-	URMS	Europa Tri-Gear	PFA 247-12922

#### 3. MODIFICATION DETAILS

Title:	Wing tip Navigation Lights & Strobes
TILIC.	Willig tip Mavigation Lights a strong

Background: describe the purpose of the mod and what it consists of.

The purpose of this modification is to install OE195 LED Wing Tip Combined Marker Lights and Strobes supplied by The Light Aircraft Company Ltd. to the wing tips of a completed and sealed Europa Classic wing.

The steps required to complete the installation have been developed so that a single hole is required in the tip of each wing to allow for power wiring and a mounting point.



# LAA/MOD 3 MODIFICATION APPLICATION Issue 5

1.	IUU	No.	
		,	

4. LIST OF ACCOMPANYING DOCUMENTS DESCRIBING MOD

Document No.	Title / Description	Issue

5. LIST OF ACCOMPANYING DOCUMENTS SUPPORTING
AIRWORTHINESS OF MODIFICATION (e.g. test report, compliance checklist, etc)

1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Issue
t Ref Title / Description	
	Title / Description

### 6. PARTS LIST

List any new manufactured or procured parts. Where a new part is designed and used it should be drawn up with sufficient dimensions to define it and make it, including any protective treatments (e.g. paint). Where one exists, the manufacturer's part number should be given, otherwise it should be allocated a part number in the following convention: Part number = M (Mod number)-(sequence number), example: M12345-01.

Qty	Part No.	Description	Source
1 pair	OE195	LED Wing Tip Combined Marker Lights and Strobes	The Light Aircraft Company Ltd.
2	-	Plastic tube	Sourced locally
2	314-1198	2 Way Cable Mount Socket Connector	RS Components
15	22TG2T14	2 CORE 14g Screen Airframe Wire	Mendelssohn Pilot Supplies
2	233-455	2.5 mm Cable Tie	RS Components
1	2TC2-2	2 Amp Breaker	LAS Aerospace Ltd.
1	228-4581	Double-pole single throw switch	RS Components
1	G015700101	RTV157 High Strength Silicone Sealant	LAS Aerospace Ltd.
2	131583838983	AMP Superseal Waterproof Electrical Connector	3 Way Components Ltd.

Attach a continuation sheet as required.



# LAA/MOD 3 MODIFICATION APPLICATION Issue 5

Mod	No.	
	Mod	Mod No.

### 7. IMPLEMENTATION

#### Describe:

- Health and safety provisions
- Any disassembly requirements
- Instructions for Installation of new parts.
- Re-assembly requirements to make aircraft flyable again.

In a suitable, well lit, dry location, chock the aircrafts wheels to prevent it from moving.

This modification applies to a Europa classic wing and LED Wing Tip Combined Marker Light and Strobes units from The Light Aircraft Company.

With the wing still attached to the aircraft, remove the aileron bell-crank inspection cover. Cut a hole approximately 2 cm by 2 cm in the side of the bell-crank housing recess wall that faces the wing-tip.

Drill a hole 6 mm in diameter through the surface of the wing tip, ??? cm from the leading edge and equidistant between the top and bottom of the wing's surface. Make the hole 50 mm deep.

Push a long metal instrument (a long screwdriver for example) through the hole towards the trailing edge of the wing at an angle of approximately 30-40 degrees until it reaches the channel that runs through the length of foam core (numbered 4 on page 7-6 of the Europa Monowheel manual dated 17th Jan 1996).

Push the strobe power wiring through the hole made in the wing-tip and into the channel. Continue pushing the strobe power wire through the wing-tip hole until it is accessible through the hole cut in the wall of the bell-crank housing recess. Pull enough wire through wing for it reach 30 cm next the root rib

With the wing removed, from the bell-crank recess, feed the wire down the aileron link-rod channel until it appears through the root rib.

Secure the wire to the wing surface covered by the wing faring by drilling a small hole through it, passing a tie-wrap through the hole and securing the wire by closing the tie-wrap around it.

Prepare the wire and fit a two-pole male AMP Superseal waterproof connector to it.

Remove the pilots seat and the two screws securing the plywood panel beneath it. Remove the plywood panel and all foam packing to expose the inside surface of the fuselage.

From inside the cockpit, drill a small hole through the fuselage approximately 1 1/2 cm in-front of the rear face of the seat back and 1 1/2 cm above the floor of the fuselage so that it appears in the area covered by the wing and in-front of the spar socket.

Prepare a length of strobe power wire and fit a waterproof two-pole female connector to the end. Feed the wire through the hole in the fuselage with sufficient length that it can be routed to the common earth bus bar behind the cockpit instrument panel, and to the master switch 12 volt bus bar via a suitable switch and breaker. Seal the hole with a suitable silicone sealer.

Returning to the wing-tip, thread the strobe power wire protruding from it through a 50 mm length of plastic tube with an outside diameter of 6 mm.

Push the plastic tube carrying the wire through the hole in the wing-tip until only 10 mm remains outside the wing-tip. This will require additional effort as the wire and tube will be compressing the foam as it is push in. This will be used as the locater tube that the strobe units attach to.

Double-check you are fitting the correct strobe to the port wing - the red strobe.

Shorten the wires attached to the strobe to approximately 5 cm and solder them to the strobe power wiring, making the wires as short as possible and insulating them with heat-shrink tubing.

Test-fit the strobe by gently pushing the wiring into the plastic locater tube and locating the hole in the strobe's base-plate over the protruding plastic tube. Be careful not to put strain on the wires soldered to the strobe's circuit board. If the strobe won't touch the surface of the wingtip without putting strain on the wires, remove the strobe and push the plastic tube further into the wing-tip and repeat the test.

Once the strobe locates correctly over the tube, clean the surface of the wing tip with a suitable cleaning product and use high strength silicone sealer to secure the strobe to the wing-tip.

Repeat for the other wing and strobe.

Re-attach the wings, connecting any pitot tubes and the strobes before locating the wings fully home and securing them in the usual way.

	Is the proposed modification is curre (Note that once installed, the aircraft ma	ntly fitted to the aircraft?	YES	NO
	If yes, briefly describe any other	iy not be nown and permission given)		
The second name of the second	work needed for the aircraft to be			
	ready for flight (e.g. finish re-build)	3		
	If no, when do you intend to fit the	As soon as the approval is receive	d.	
	modification?	70 00011 do 1110 app. 01 d. 10 10 10 11		



# LAA/MOD 3 MODIFICATION APPLICATION

Mod	No.	
		-

Issue 5

8. WEIGHT AND BALANCE EFFECT ON AIRCRAFT

OI AAFTOILL LAL	THE BUTTON		44014040	
Date of current W&B report: 14th April 2016		Weight (lb/kg)	CG (in/mm)	Moment
	A/C pre-mod	913 lbs	60.47	<i>55205</i>
+/- weight change		+0.1 lbs	(moment arm) <b>58</b>	5.8
= A	/C post-mod	913 lbs	60.47	55211

## 9. INSPECTION CHECKS AND SPECIAL INSTRUCTIONS

All modifications will need to be inspected for quality and conformity. State any additional proposed requirements e.g. function check, full and free checks, friction, clearance from other structures and systems, etc.

A functional	check	of providing	power to	the	strobes	will	be	made	prior	to	each
flight.											

### 10. FLIGHT TEST AND SPECIAL INSTRUCTIONS

Example statements may be:

- 25 hours of reliability testing.
- Full performance and handling check by a qualified test pilot.
- 20 landings on various surfaces.

None

#### 11. OWNER'S DECLARATION

I declare that the foregoing information is correct and I agree to abide by any conditions pertaining to this modification.  I agree that this modification, if approved, can be used free of charge by others.		
Name (owner): (on behalf of all the owners)	Carl Parkinson	
Signature:	///	
Date:	6th Feb 2017	

Note: a signature is not required if the owner is submitting this form by email; however, by submitting the application, you signify that you agree with the Owner's Declaration.

If this mod is successful, are you willing to allow potential applicants wishing to fit the same mod to their aircraft to contact you? YES / NO

If so, which means of contact is acceptable to be published on the LAA's web site? home phone  $\Box$ , mobile phone  $\Box$ , email  $\blacksquare$ , address  $\Box$  [Tick whichever apply]